# SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM : SEPARATION MECHANISMS-PYRO FMEA NO F2-3A -F3 -1 REV:10/09/87

ASSEMBLY :FORWARD SEPARATION BOLT

CRIT. FUNC: :SKD26100098-245 CRIT.

P/N RI P/N VENDOR:

VEHICLE 102 103 104 QUANTITY :1 EFFECTIVITY: Х х X

PHASE(S): PL X LO X OO 00 LS

HDW:

REDUNDANCY SCREEN: A-

PREPARED BY: APPROVED BY: 10/1/87APPROVED BY (NASA): DES R. H. YEE + TUGRANES

REL MACHENIA OFFICE SSM RUN REL M. B. MOSKOWITZ Harried and QΕ E. M. GUTIERREZ QΕ 

ITEM:

FORWARD ATTACH SHEAR BOLT

### FUNCTION:

STRUCTURALLY TIES TOGETHER THE ORBITER/EXTERNAL TANK (ET) AT FORWARD ATTACH POINT. FRACTURES UPON RECEIVING A PRESSURE OUTPUT FROM EITHER OR BOTH CARTRIDGES.

### FAILURE MODE:

PREMATURE BOLT FRACTURE

## CAUSE(5):

IMPROPER MACHINING AT SEPARATION AREA, MATERIAL DEFECT, INTERNAL CORROSION, EXCESSIVE PRELOAD/LOW PRELOAD, EXCESSIVE PRICTION IN MULTIPLECE SPHERICAL BEARING (FMEA/CIL 02-3A-F4-1), PREMATURE PYRO FIRING (FMEA/CIL P2-3A-F1-2)

## EFFECT(S) ON:

(A)SUBSYSTEM (B)INTERFACES (C)MISSION (D)CREW/VEHICLE

(A,B,C,D) LOSS OF FUNCTION - POSSIBLE LOSS OF CREW/VEHICLE DUE TO LOSS OF ORBITER/ET STRUCTURAL INTEGRITY.

#### DISPOSITION & RATIONALE:

(A)DESIGN\_(B)TEST (C)INSPECTION (D)FAILURE HISTORY (E)OPERATIONAL USE

## (A) DESIGN

SAFETY FACTOR EQUAL OR GREATER THAN 1.4. BREAKOUT TORQUE OF 100 INCH-LB MAXIMUM WHILE IN MULTIPIECE SPHERICAL BEARING, BOLT SHANK STRAIN GAUGES PROVIDE ACCURATE PRELOAD READOUT DURING ORBITER/EXTERNAL TANK MATE. BOLT IS INCONEL 718 FOR CORROSION PROTECTION (ULTIMATE TENSILE 180-200 KSI).

## (B) TEST

COMPONENT QUALIFICATION TESTS: TENSION/SHEAR-LIMIT, ULTIMATE LOADS, AND CERTIFICATION REQUIREMENTS (CR) 45-325-0014.

SYSTEM QUALIFICATION TESTS: 9 FIRINGS AT AMBIENT (6 UNDER LOAD), STATIC LIMIT AND ULTIMATE LOADS TESTS. CR-45-562001.

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ACCEPTANCE TESTS: 100% DYE PENETRANT, 100% ULTRASONIC TEST ON SHANKS AND HOUSINGS, 100% HOUSINGS HYDROSTATICALLY PROOF PRESSURE TESTED. 100% SHANK HARDNESS TEST, 100% SHANK PROOF LOAD, TENSILE TEST COUPONS FOR HOUSING, FISTON, AND SHANKS VERIFIES MATERIAL INTEGRITY. PROOF TEST ACCEPTANCE CRITERIA IS NO EVIDENCE OF CRACKS, DEFORMATION, OR PERMANENT SET. CR-45-325-0014, ATP 8664; SKD26100098, ATP 8875.

OMRSD: TURNAROUND TESTS INCLUDE - SPHERICAL BEARING ASSEMBLY BREAKOUT TORQUE VERIFICATION AFTER EACH FLIGHT, BOLT/BEARING ASSEMBLY VERIFICATION FOR NO BINDING DURING BUILD-UP, POST-TORQUE EXPOSED THREAD CHECK, FORWARD SEPARATION BOLT/NUT THREAD INSPECTION, AND BOLT PRELOAD TORQUE VERIFICATION AFTER 12 HOUR HOLD PER MLO302-0016. NEW HARDWARE INSTALLED EACH FLIGHT.

## ) INSPECTION

## RECEIVING INSPECTION

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RAW MATERIAL IS VERIFIED BY INSPECTION TO ASSURE SPECIFIC SHUTTLE REQUIREMENTS ARE SATISFIED.

## CONTAMINATION CONTROL

CONTAMINATION CONTROL AND CORRUSION PROTECTION PROCESSES VERIFIED BY INSPECTION.

## ASSEMBLY/INSTALLATION

SHANKS 100% DIMENSIONALLY VERIFIED AT SEPARATION CROSS SECTION. SELECTED MANUFACTURING/ASSEMBLY STEPS ARE IDENTIFIED BY NASA AND QUALITY ASSURANCE AND VERIFIED BY GOVERNMENT INSPECTION MANDATORY INSPECTION FOINTS (MIFS).

# MONDESTRUCTIVE EVALUATION

ONE HUNDRED PERCENT VISUAL INSPECTION, IDENTIFICATION PERFORMED, PARTS PROTECTION VERIFIED BY INSPECTION.

# RITICAL PROCESSES

ALL MANUFACTURING PROCESSES SUCH AS WELDING, PLATING, HEAT TREATING, PASSIVATION AND ANODIZING ARE VERIFIED BY INSPECTION.

### TORAGE

STORAGE ENVIRONMENT VERIFIED BY INSPECTION.

FAILURE HISTORY

OPERATIONAL USE